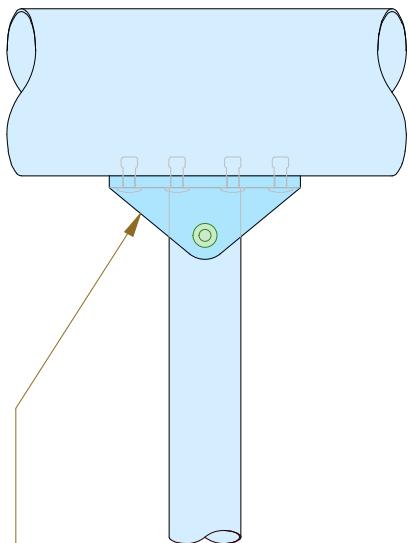
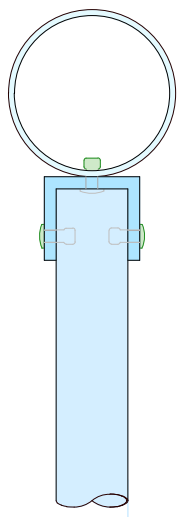


VIEW C FROM MAIN  
DRAWING AT RIGHT



COMPRESSION STRUT  
FIXTURE, FROM INSET AT  
RIGHT, ONE AT EACH END  
OF COMPRESSION STRUT,  
FASTEN WITH RIVETS

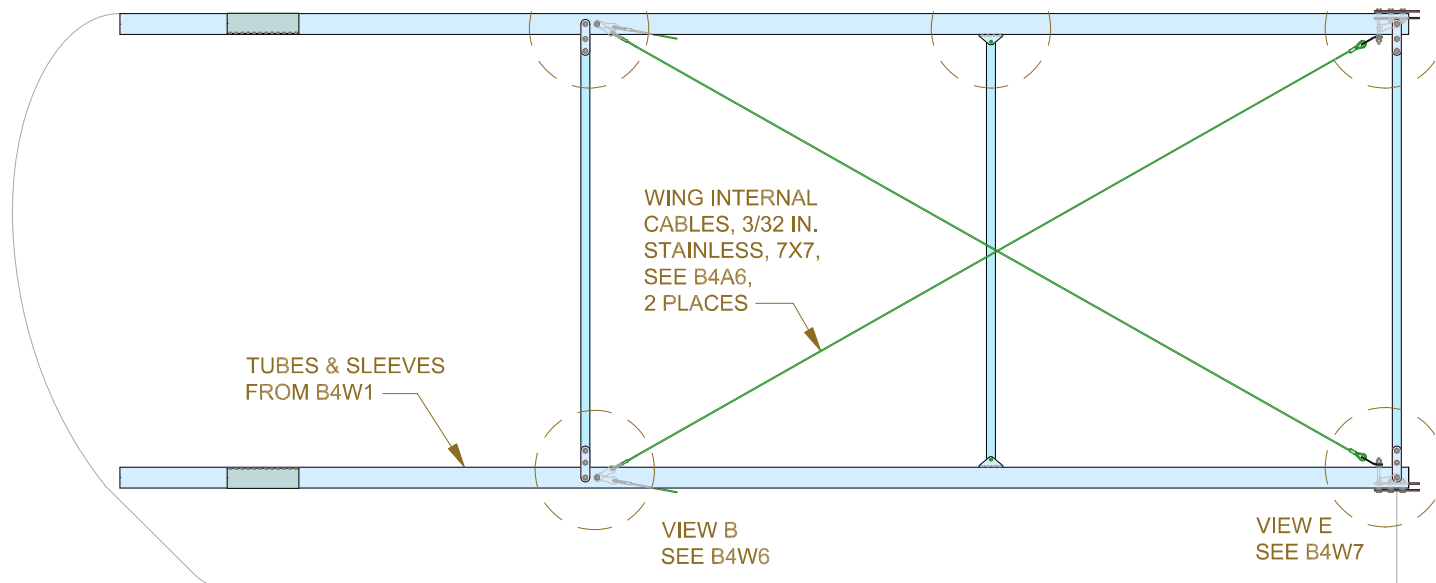


SAME AREA AS VIEW C ABOVE,  
VIEWED FROM OUTBOARD

VIEW A  
SEE B4W6

VIEW C, SEE  
INSET AT LEFT

VIEW D  
SEE B4W7



TUBES & SLEEVES  
FROM B4W1

WING INTERNAL  
CABLES, 3/32 IN.  
STAINLESS, 7X7,  
SEE B4A6,  
2 PLACES

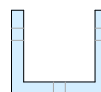
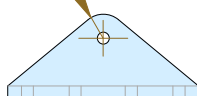
VIEW B  
SEE B4W6

VIEW E  
SEE B4W7

OUTBOARD UPPER WING FRAME ASSEMBLY, UPPER LEFT WING VIEWED FROM  
ABOVE, RIGHT & LEFT WING PANELS ARE SIMILAR (MIRROR IMAGE LEFT & RIGHT)

HARDWARE & INSTALLATIONS IN VIEWS A,B, & C ARE THE SAME FOR THE  
FRONT & REAR SPAR, BUT ARE REVERSED FORE & AFT (MIRROR IMAGES)

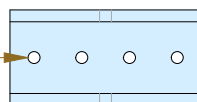
1/4 R



1/2 TYP.



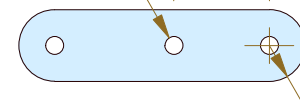
1/8 DIAM. HOLE,  
6 PLACES



1 REF.

COMPRESSION STRUT FIXTURE, 1 X 1/8 X 7/8  
ALUM. CHANNEL, 2 IN. LONG, MAKE 8,  
SEE INSET AT LEFT

3/16 DIAM. HOLE,  
3 PLACES



3/8R, BOTH ENDS

STRUT FIXTURE, 3/4 X 1/8  
ALUM. BAR, 3 IN. LONG,  
MAKE 34, SEE B4W4

B4W5

OUTBOARD  
UPPER WING  
FRAME  
ASSEMBLY

BLOOP 4  
ULTRALIGHT  
AIRPLANE

M SANDLIN  
AUGUST 21  
2016